

Appln. No. 10/823,016  
Reply to Office Action of Sep. 20, 2005

**Amendments to the Specification:**

**Please replace paragraph beginning on page 1, line 31 with the following amended paragraph:**

According to a particular object of the present invention, the stationary blades, i.e., the nozzles of such a high-pressure module of a steam turbine are of a profile provided with channels that converge and then diverge going from the inlet to the outlet of such a high-pressure module. This profile for the stationary blades, which may also be known as "nozzles", is established on the basis of supersonic flow theory. Each such ~~moving-stationary~~ blade has a profile such that it limits separation and losses along the channels, and this profile is established by using complex calculations in three dimensions and by using aerodynamics equations.

**Please add the following new paragraph after the paragraph ending on page 4, line 3:**

The stationary blades, i.e., the nozzles 24 of the high-pressure module 2, are of a profile provided with channels 26 that converge and then diverge going from the inlet to the outlet of the high-pressure module 2. This profile for the nozzles 24 is established on the basis of supersonic flow theory.

**Please replace paragraph beginning on page 5, line 4 with the following amended paragraph:**

The spiral shape of the inlet volute 22 makes it possible to generate a uniform flow at the inlet end of the supersonic nozzle 24 at all azimuth angles.

**Please replace paragraph beginning on page 5, line 6 with the following amended paragraph:**

The supersonic nozzle 24 may be a multi-channel nozzle. The steam can then enter the body via a multitude of openings.